

# **Pegasus 5: An Automated Pre-Processor for Overset-Grid CFD**

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Overset Short Course

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# Acknowledgements

- **Pegasus 5 Primary Authors:**  
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- **Contributing Authors:**  
Steve Nash, William Chan, Robert Tramel, Jeff Onufer
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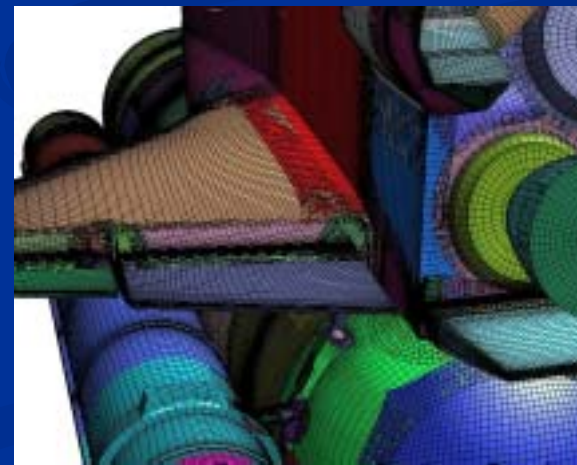
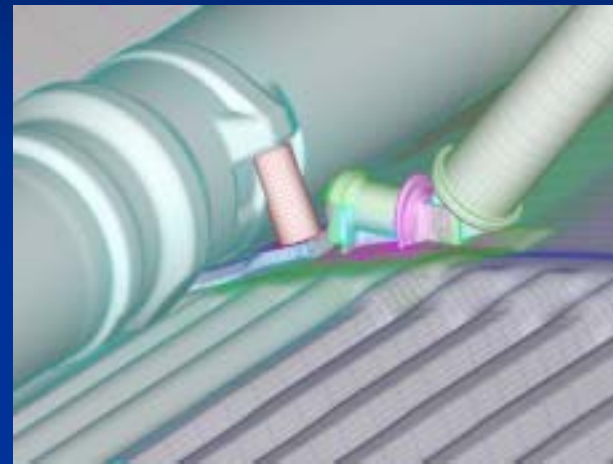
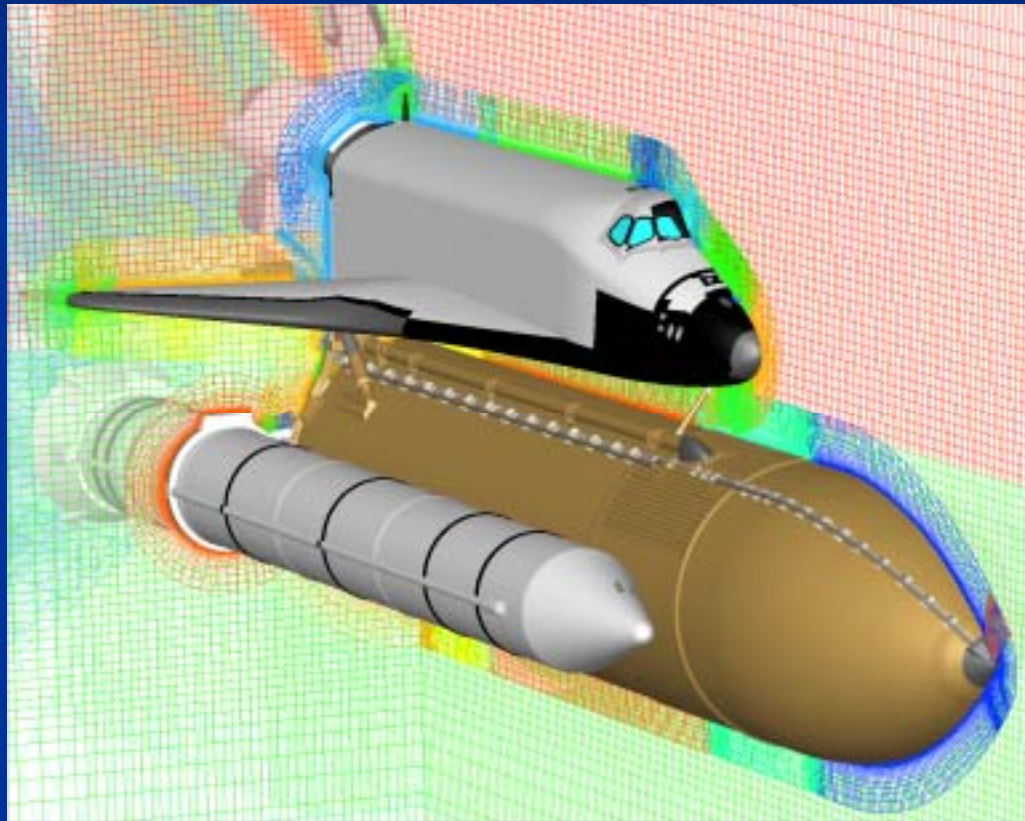


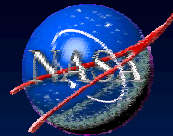
# Outline

- Understanding Overset-grid work flow
- Pegasus5 features and automation
  - Auto hole cutting
  - Interpolation and overlap optimization
  - Projection
  - Restarting
  - Parallelization
- Overview of Usage
  - Required input
  - Basic Usage
  - Understanding the output
  - Overcoming problems

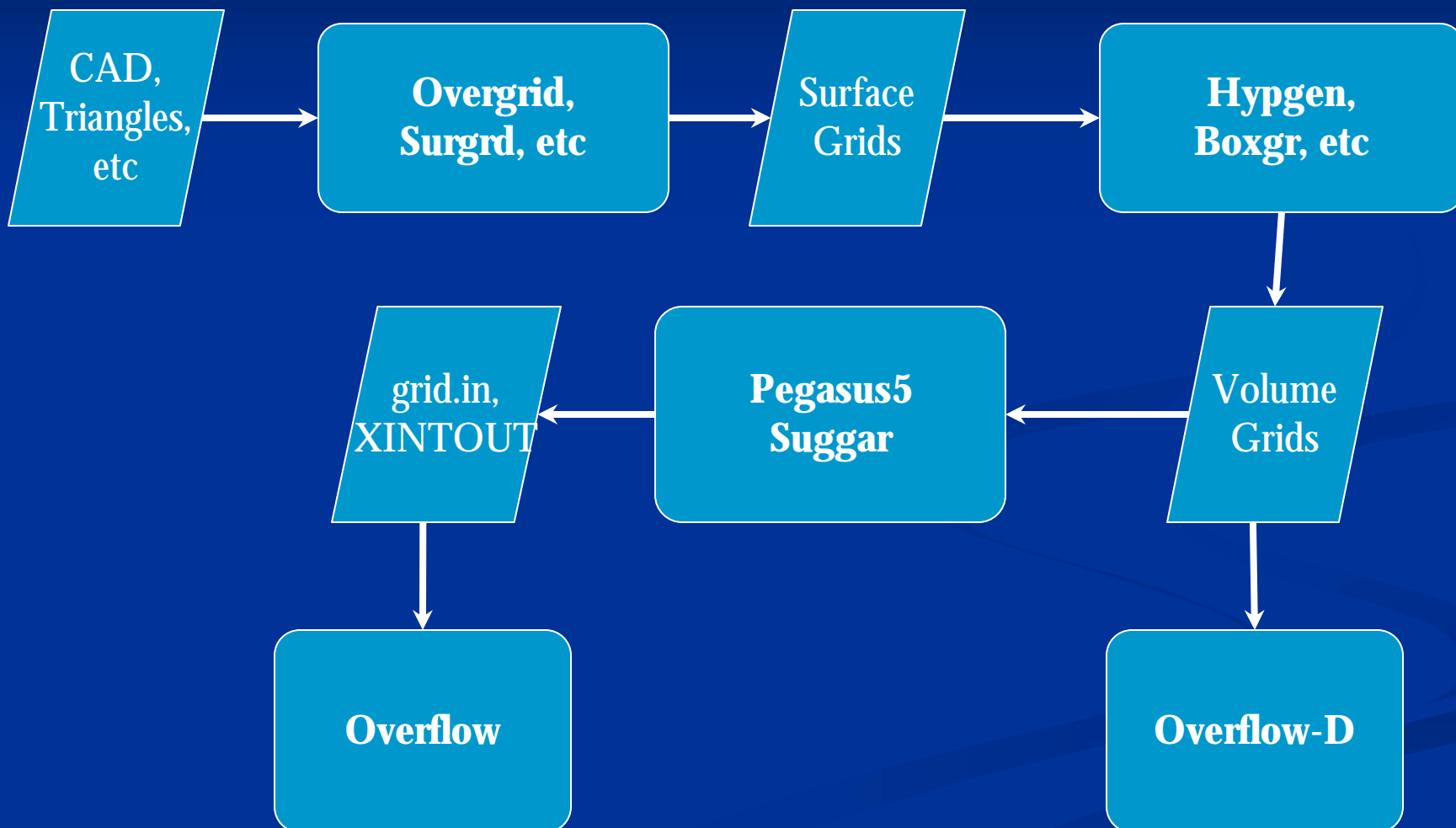


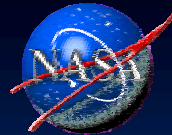
# The Oversetting Challenge





# Overset-CFD Workflow





# Pegasus5 Goals and Features

- Fifth-generation overset software
  - Written in 1998-2000 as a replacement for *PEGSUS4*
- Primary goal: complete automation of overset process
  - Complexity of CFD problems continues to grow
  - Hundreds of overset zones, tens of millions of grid points
  - Manual control of process became intractable
- Required all-new approach to:
  - Hole-cutting
  - Overset optimization
- Required significant improvements in ease of use
  - Parallelization
  - Automatic restarts
  - Projection
- Maintained many *PEGSUS4* features, allowing manual control where needed
- Pegasus5 is mostly automated compared to previous generation, but still requires user knowledge and expertise

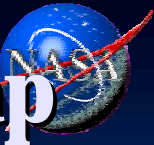


# Pegasus5 Approach

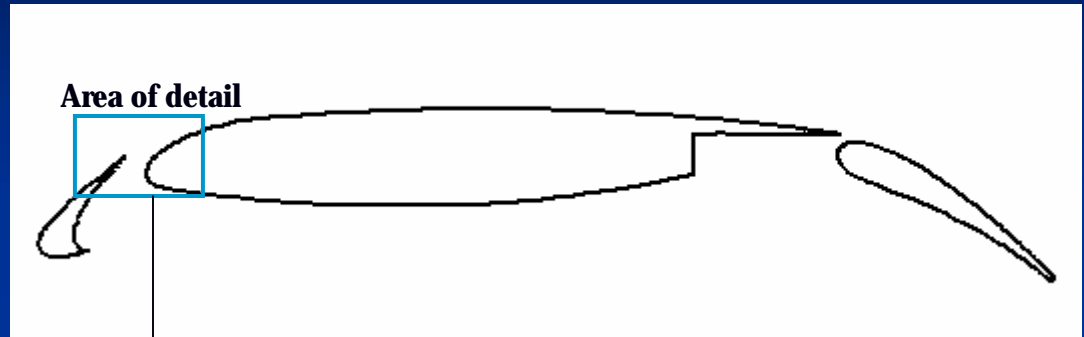
- Use an Overflow-like namelist input file
  - Parsing the flow-solver boundary conditions provides most of required inputs about geometry and topology
- Use Fortran90
  - Extensive use of defined-type data and modules
  - Extensive use of process templates and data templates
- Oversetting task broken into discrete tasks
  - Input to each task saved as one or more files
  - Result of each task saved as one or more files
  - Facilitates parallelization
  - Enables restarts from partial or aborted run
  - Enables rapid restarts after change to inputs
- Use lots of computer time and lots of disk I/O
  - Intelligent use of brute-force can solve lots of issues

# Auto-Hole Cutting Uses a Cartesian Map

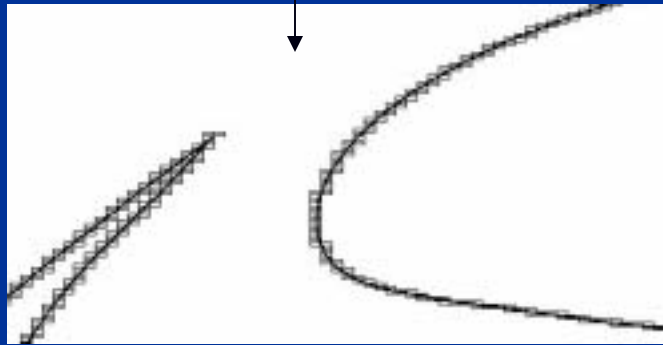
## Example: Multi-element Airfoil



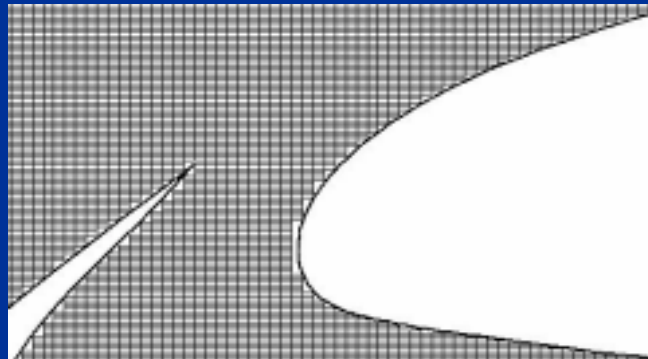
1. Identify the airtight solid-wall surfaces and overlay with a Cartesian map



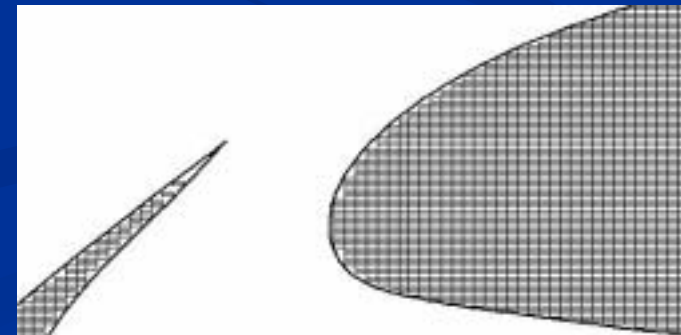
2. Find **Fringe elements**: any Cartesian element which intersects the solid wall



3. Use painting algorithm to identify **Outside elements**



4. All others are **Inside Elements**

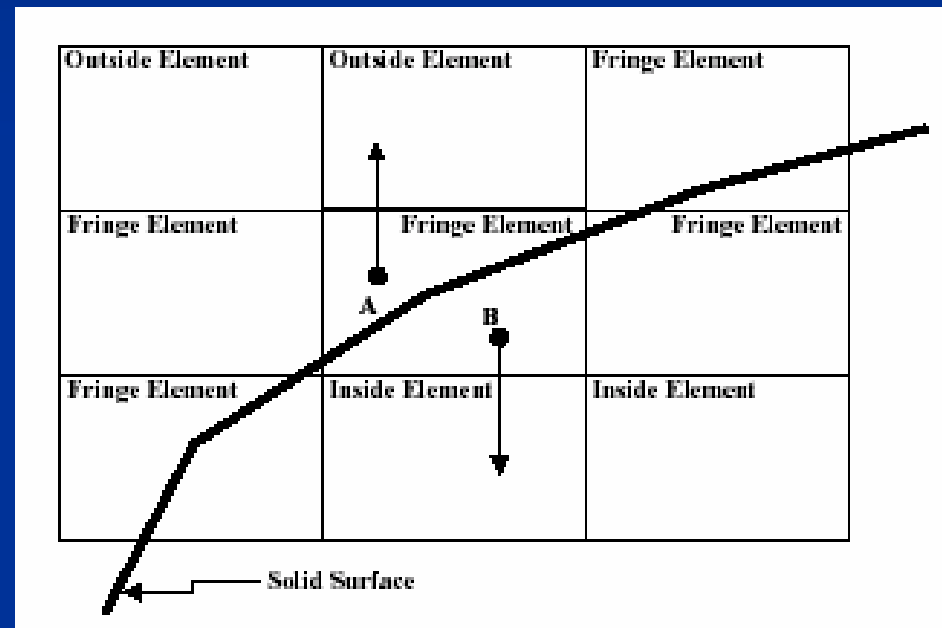


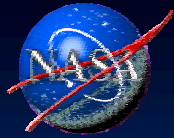


# Auto Hole Cutting: Cutting of Candidate Points

All volume grid points are tested as potential hole points:

- Points outside the Cartesian map are not hole points
- Points contained in an **Outside Element** are not hole points
- Points contained in an **Inside Element** are hole points
- Points contained in a **Fringe Element** undergo line-of-sight test
  - Point A can “see” an Outside Element without crossing the solid surface: it is not a hole point
  - Point B can “see” an Inside Element without crossing the solid surface: it is a hole point





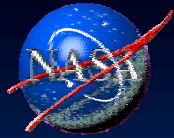
# Interpolation Stencil Search

- Pegasus5 searches for all possible interpolation stencil donors from all zones **for every single grid point**
  - Alternating Digital Tree (ADT) is created for all zones
  - For a given grid point and a given donor zone, an ADT lookup provides a near-by cell in donor zone, then a stencil-jumping approach finds the exact donor cell and interpolation stencil
  - All possible donor cells for every single grid point are stored



# Fringe Point Identification

- A fringe point is one which requires updating in the flow-solver via interpolation from a neighboring zone
- Outer-boundary fringe points
  - All points on the boundary of a zone that do not receive a flow-solver boundary condition is identified as an outer-boundary fringe point
  - Single or double outer-boundary fringes can be requested
- Hole-boundary fringe points
  - Points adjacent to a hole point are identified as hole-boundary fringe points
  - Single or double hole-boundary fringes can be requested



# Level-1 Interpolation

- For each fringe point, the best possible interpolation stencil is chosen amongst all valid donor cells
  - When multiple donors are available, selection is based on measure of interpolation quality and relative cell size
- Any fringe point which does not have a valid donor is denoted as an orphan point



# Level-2 Interpolation

- Optimization of overlap
- Interpolation points added after to Level-1 Interpolation
- Has effect of expanding the automatically-cut holes and shrinking the outer edges of overlapping zones
- Finest grid points remain active interior points
- Coarser grid points are interpolated from available donor cells of finer neighboring zones
- Methodology is robust, requires no user inputs, and maximizes communication between overlapping zones

# Level-2 Interpolation

## One Dimensional Example



Mesh A



Mesh B



Mesh C

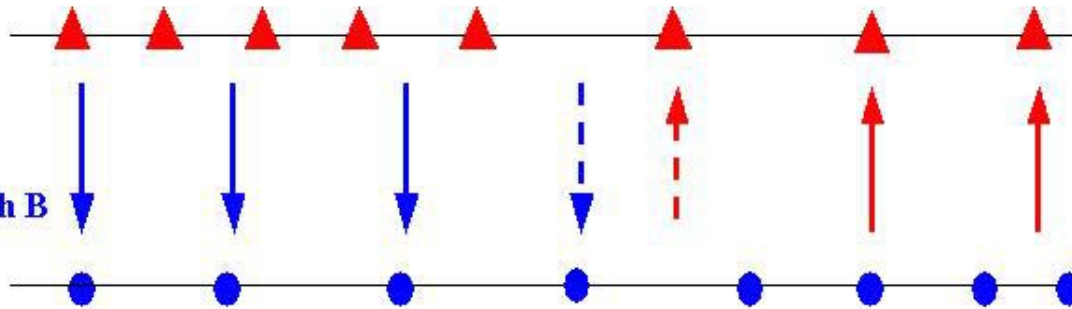


# Step 1: Interpolate Between Meshes



Mesh A

Mesh B

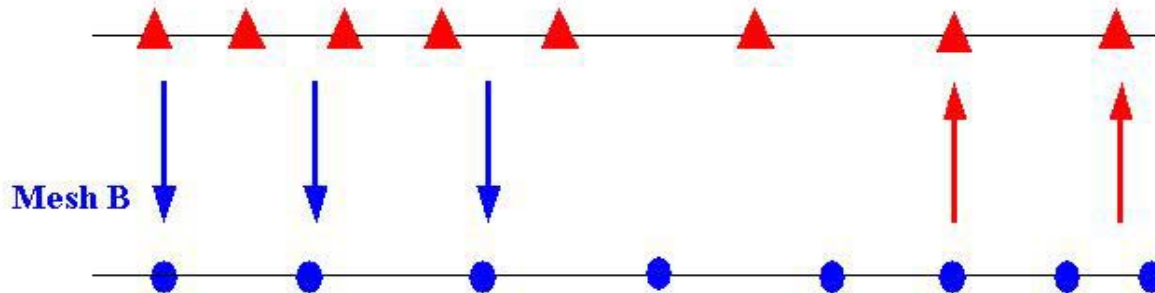


Arrow denotes direction of information flow

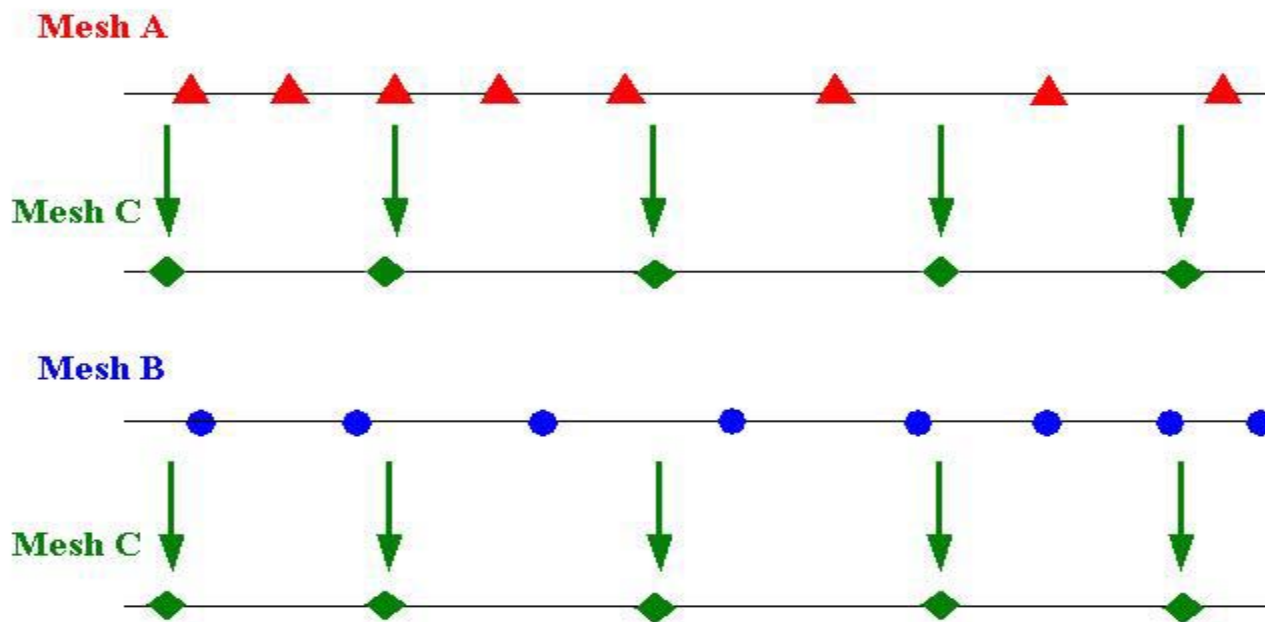
# Step 2: Remove Invalid Interpolations



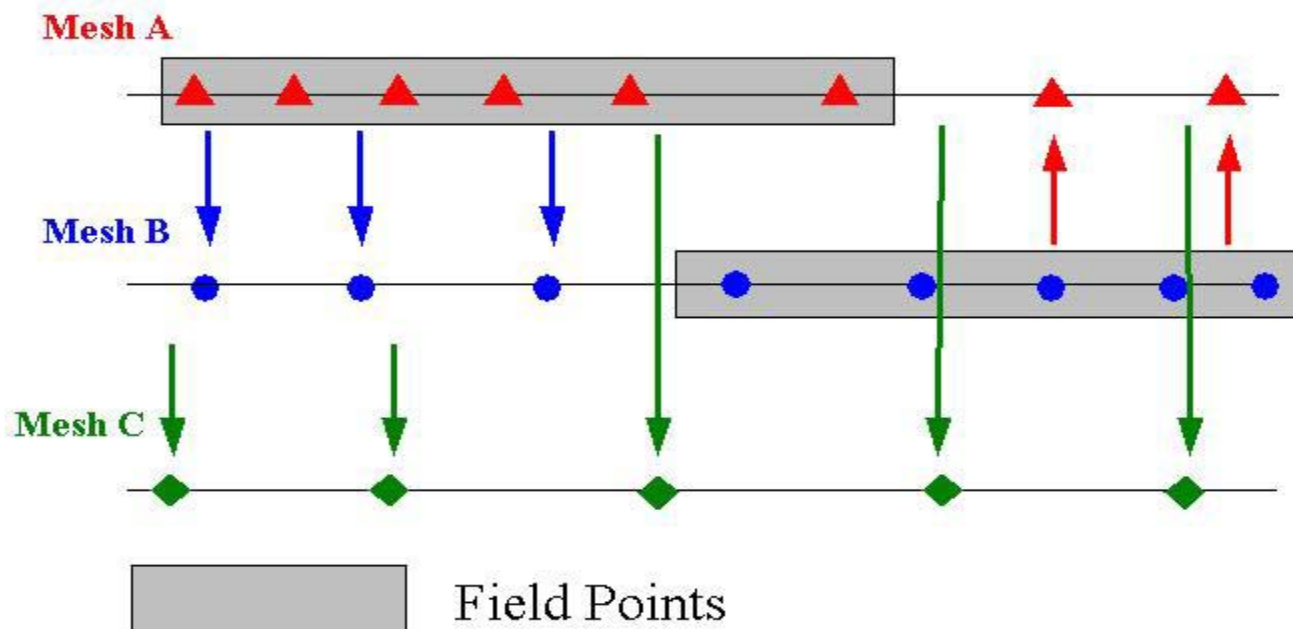
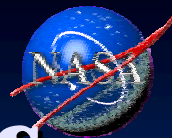
**Mesh A**



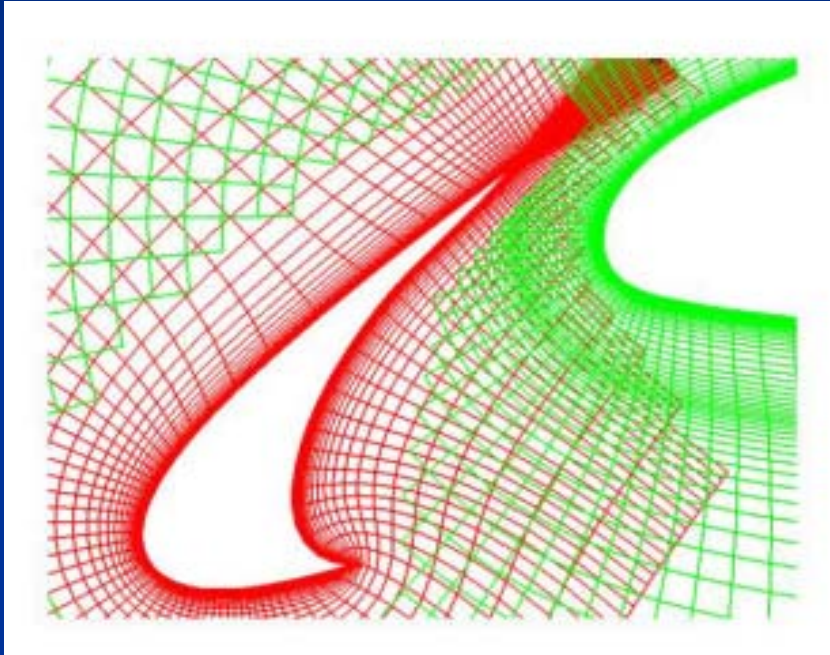
# Repeat Step 1 and 2 for Other Meshes



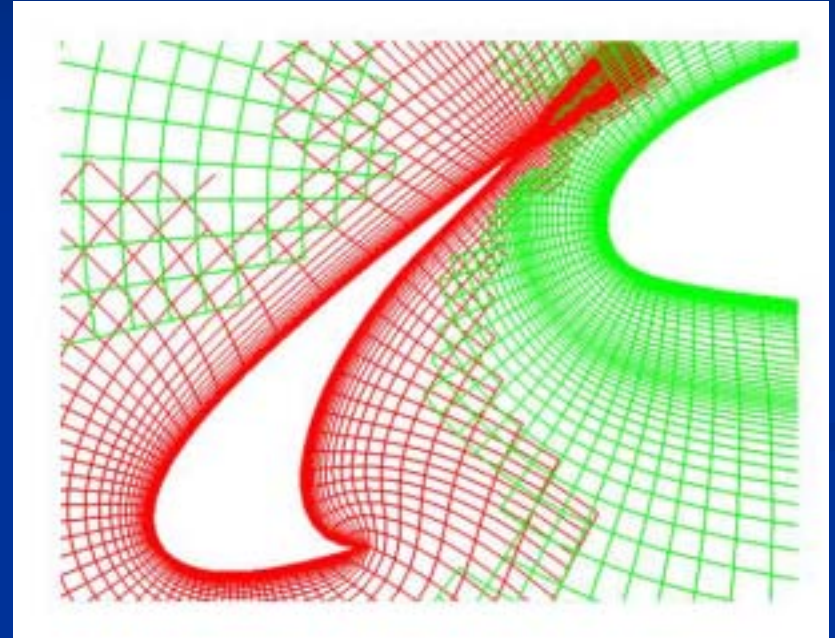
# Step 3: Keep Finest Mesh Points



# Optimized Overlap Example

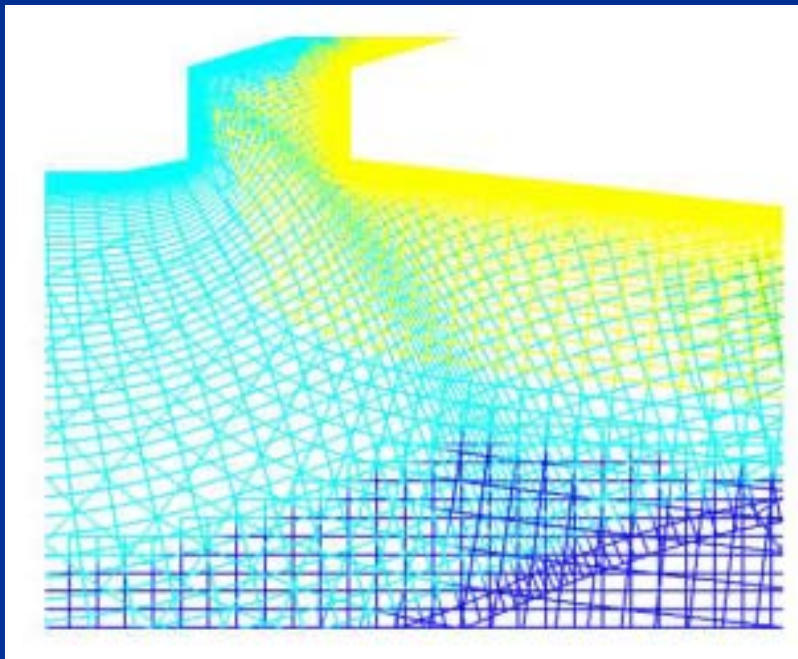


Non-Optimized Overlap

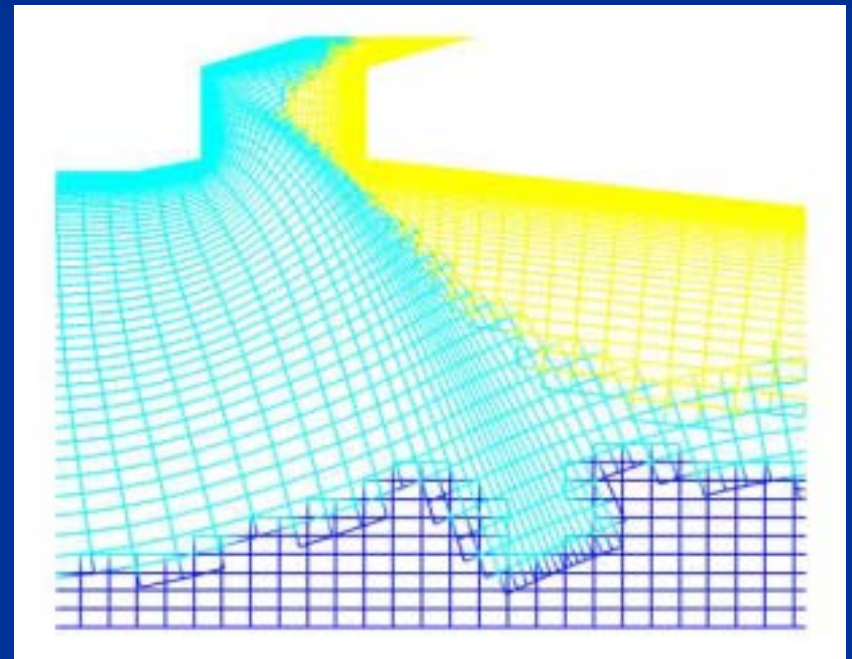


Optimized Overlap

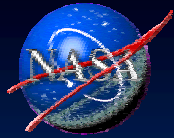
# Optimized Overlap Example



Non-Optimized Overlap



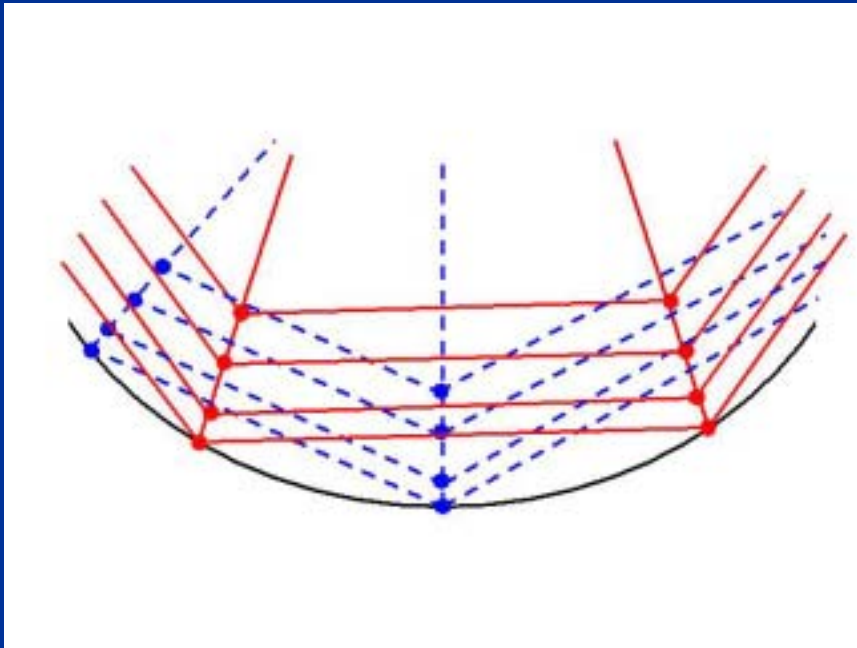
Optimized Overlap



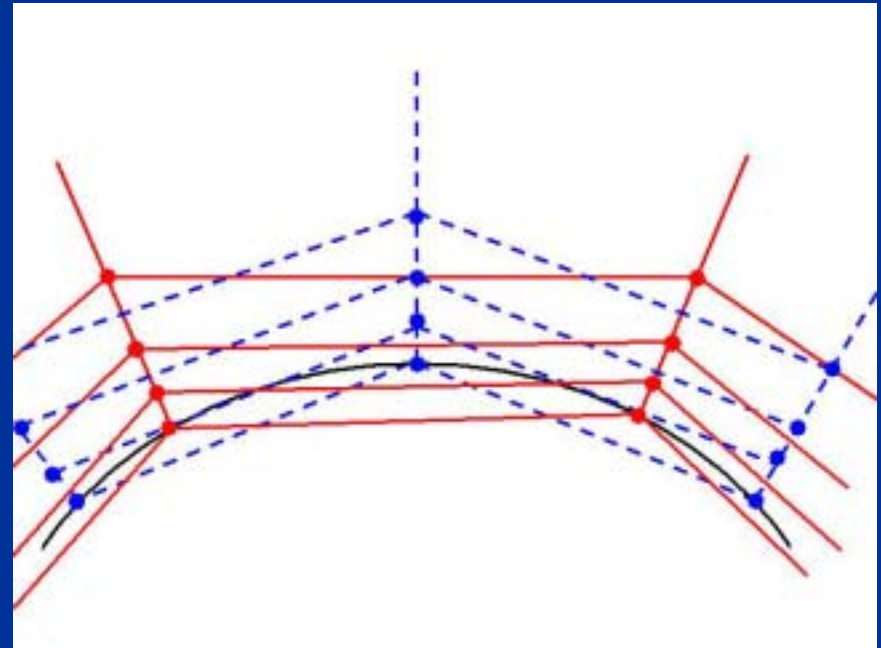
# Projection

- Corrects interpolation problems that may occur on curved viscous surfaces
  - Cell aspect ratio  $> 1000$  near viscous surface is typical
- Pegasus 5 projection step alters interpolation coefficients, not actual grid points
- Projection is performed internally and typically requires no user input

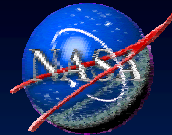
# Problem: Linear Discretization on Curved Surfaces



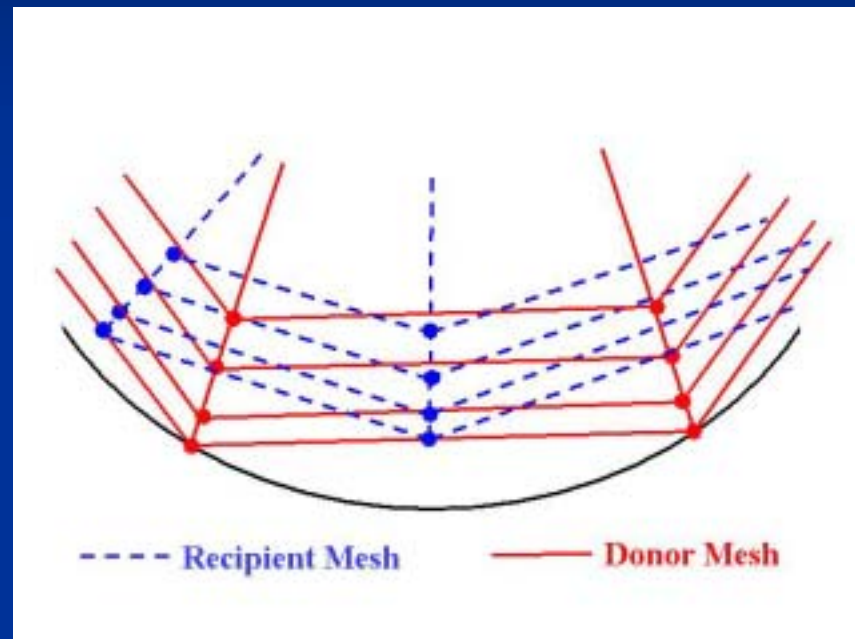
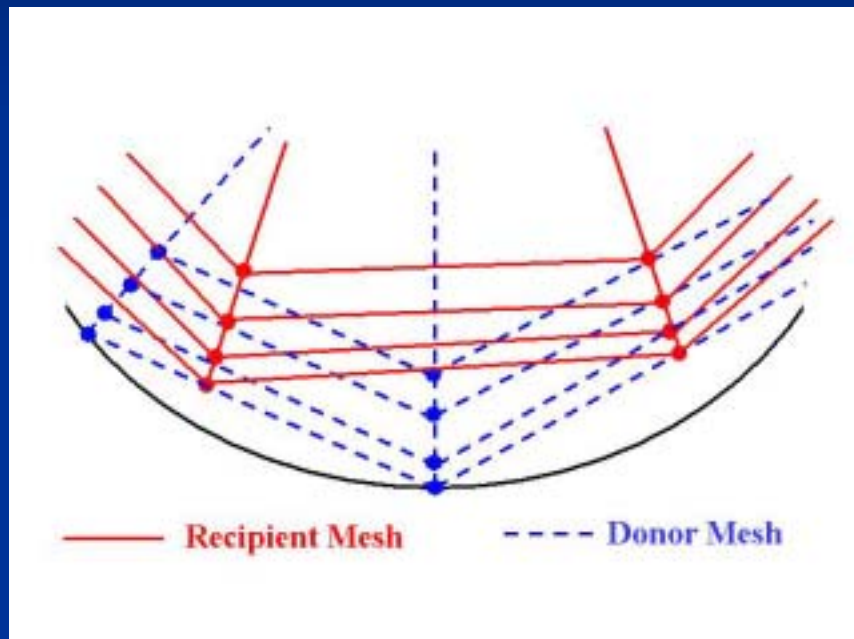
Concave Surface



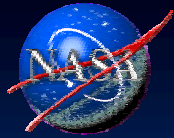
Convex Surface



# Solution: Projection



Points are Projected for Interpolation Only  
Original Meshes are Retained



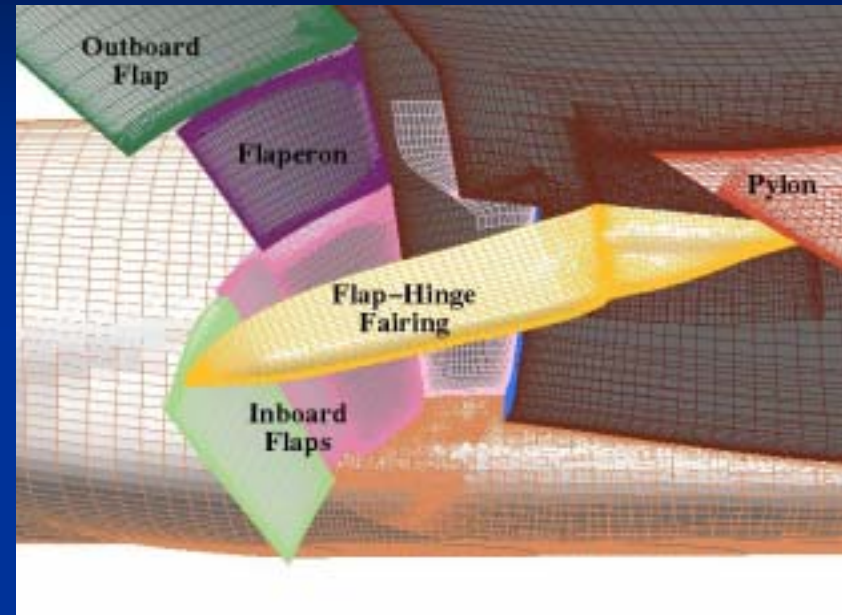
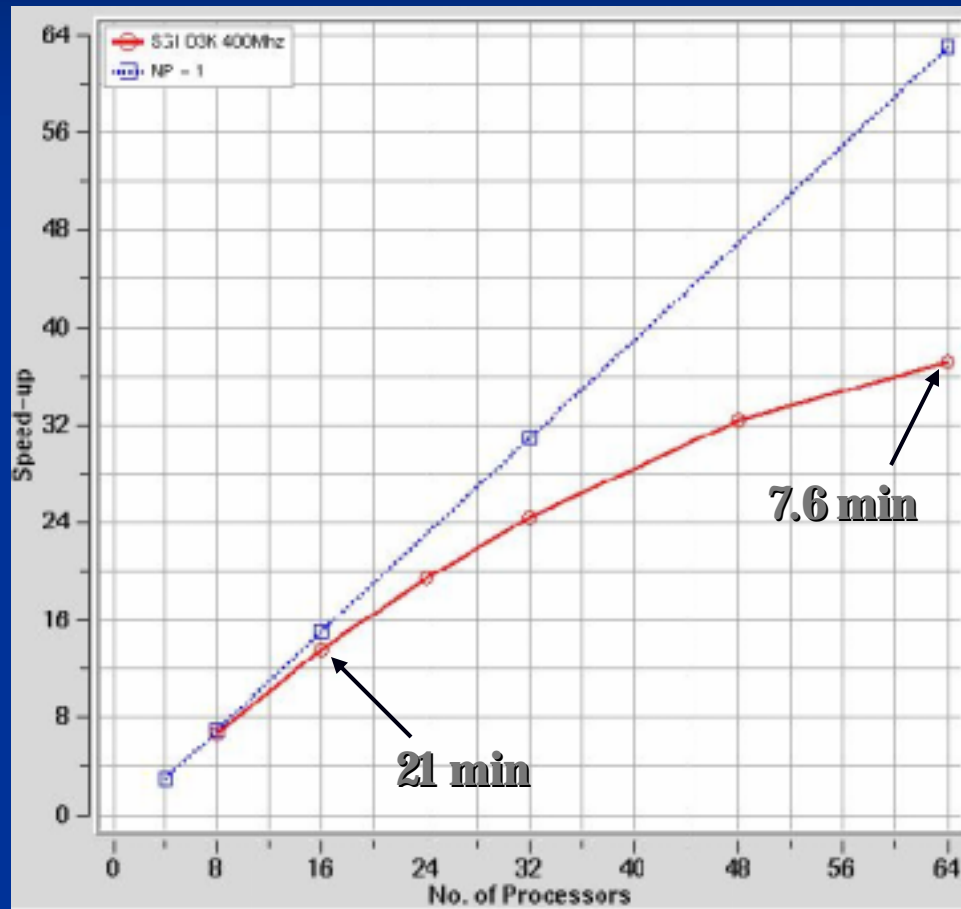
# Parallelization

- Code is composed of many tasks
  - Projection, ADT, interpolation, hole-cutting, level-1 interpolation, level-2 interpolation, etc
  - Most tasks are independent of each other
  - Each task reads its input from disk files and write their results to disk files
- Parallelization uses Message-Passing-Interface (MPI)
  - One master process to monitor and distribute the work
  - Many worker processes, one per CPU
- Reliably reproduces results of serial code
- The larger the grid system, the better the parallel scaling

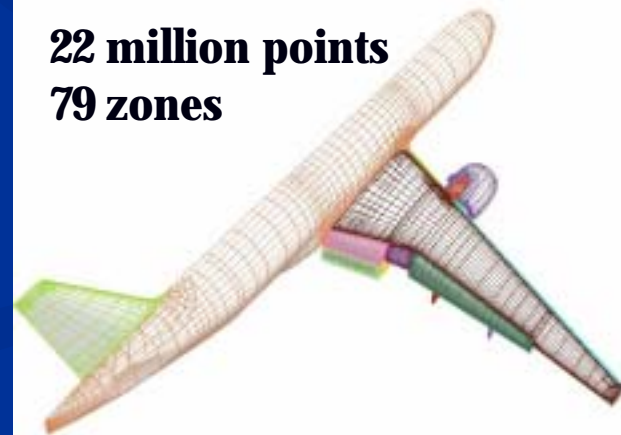
# Parallel Speedup: Boeing 777-200 Example



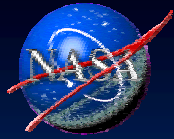
SGI Origin: Total CPU time 283 min



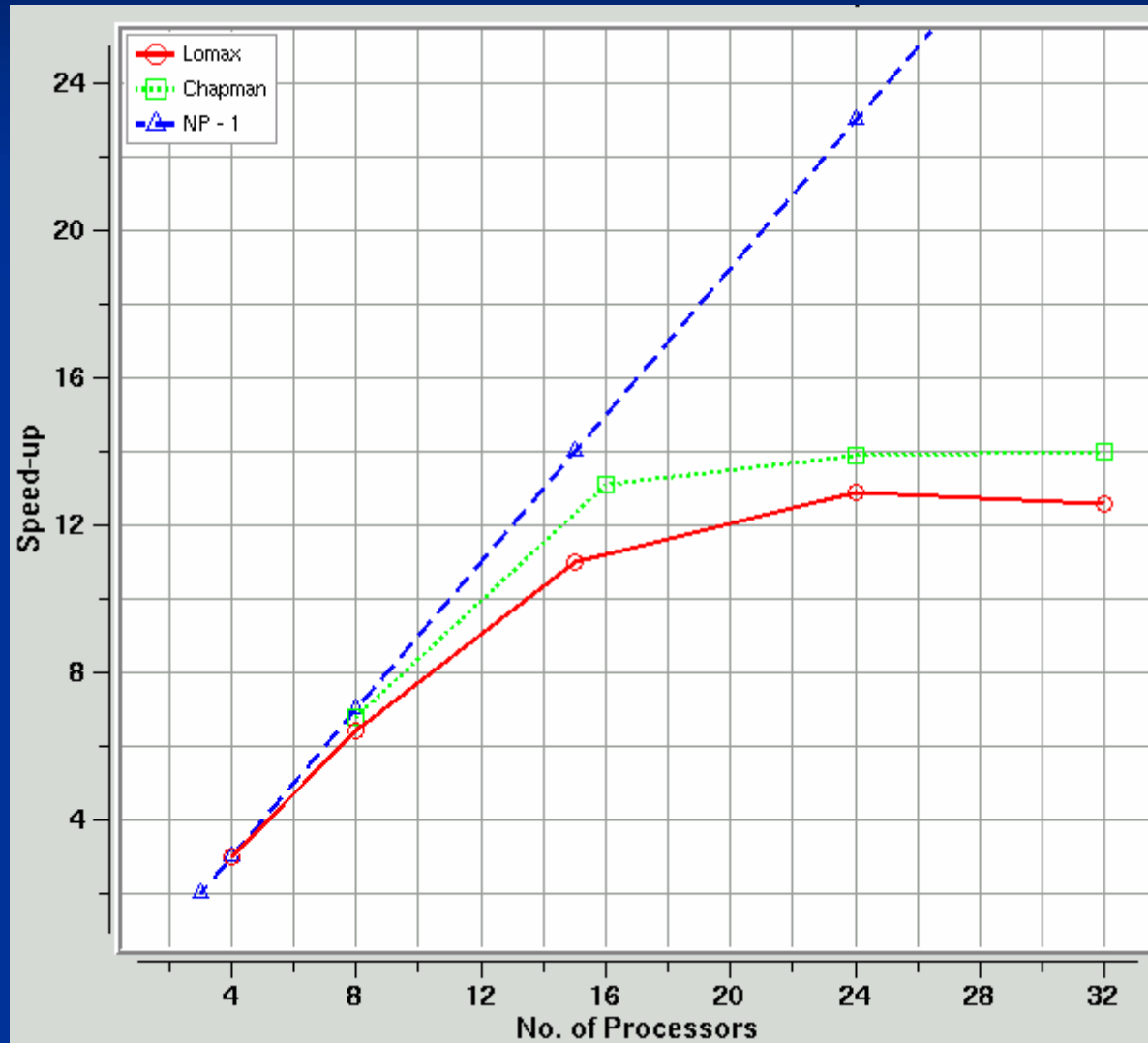
22 million points  
79 zones



# PEGASUS5 Parallel Scaling



Harrier grid system: 52 zones, 2.5 million grid points

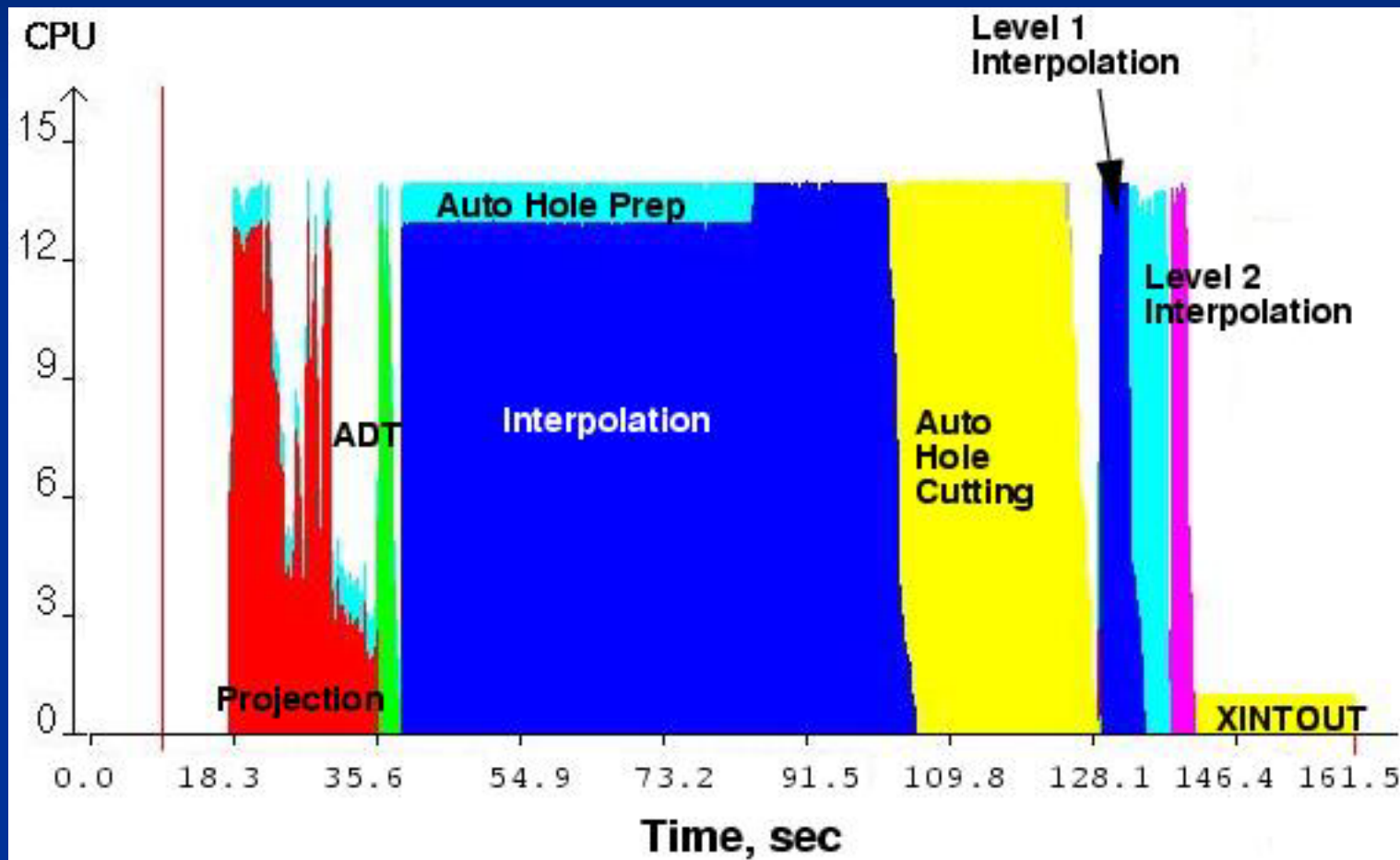




# PEGASUS 5 Parallelization

15 Processors on an SGI O2K

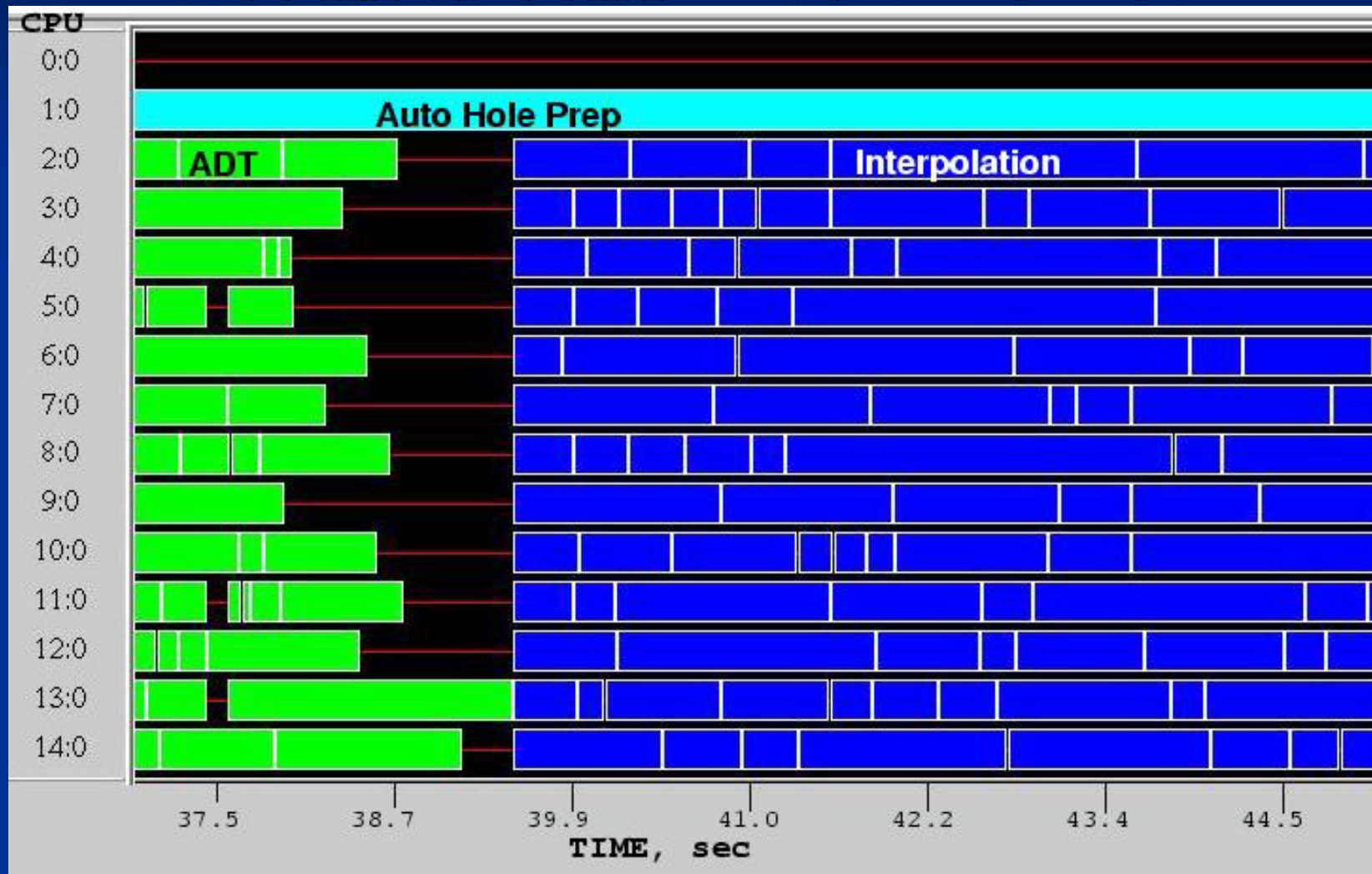
Harrier grid system: 52 zones, 2.5 million grid points

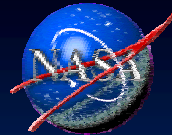




# PEGASUS 5 Parallelization

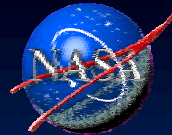
Parallel execution: barrier between ADT and INTERPOLATION





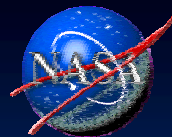
# Restarting

- Pegasus5 execution consists of many individual tasks
- Each task has a defined set of dependencies (inputs) that are stored to a disk file
- Each task results in one or more output files stored to a disk file
- Automatically determines which tasks are out of date based on internal time-stamps of input and output disk files
  - Internal time-stamps written as first and last record of each disk file
  - An incomplete or inconsistent file is considered out of date
- Upon execution pegasus5 firsts checks all files and determines which tasks need to be run
- Can successfully restart for:
  - Modifications to user inputs or zones
  - Addition of new zones
  - Incomplete previous run or computer crash
- Allows incremental buildup of complex configurations



# Pegasus5 Inputs

- Input requirements:
  - Standard input file, namelist format
  - Volume grids in individual files:  
X\_DIR/zonename1.x, .... , X\_DIR/zonenameN.x
- Tools to assist in generating these inputs
  - peg\_setup script
    - Requires Overflow input file and multi-zone plot3d grid file containing all of the volume grids
  - Chimera Grid Tools scripts: BuildPeg5i



# Pegasus5 Input File Example

```
$GLOBAL
```

```
FRINGE = 2,
```

```
OFFSET = 1,
```

```
$END
```

Set double fringe

OFFSET used to expand auto hole

```
$MESH NAME = 'body', KINCLUDE= 2, -2, LINCLUDE= 2, -1  
OFFSET=2, $END
```

```
$MESH NAME = 'bodynose', JINCLUDE= 2, -1, LINCLUDE= 2, -1, $END
```

```
$MESH NAME = 'wing', $END
```

```
$MESH NAME = 'wingcap', $END
```

```
$MESH NAME = 'wingcol', $END
```

```
.  
. .  
.
```

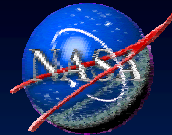


# Pegasus5 Input File Example

```
$BCINP I SPARTOF = ' body' ,  
  I BTYP = 5, 17, 17, 15,  
  I BDIR = 3, 2, -2, -1,  
  JBCS = 1, 1, 1, -1,  
  JBCE = -1, -1, -1, -1,  
  KBCS = 1, 1, -1, 1,  
  KBCE = -1, 1, -1, -1,  
  LBCS = 1, 1, 1, 1,  
  LBCE = 1, -1, -1, -1,  
  YSYM = 1,  
 $END
```

4 BCs (columns)  
5 = viscous wall  
17 = symmetry  
15 = axis

Symmetry in Y



# Pegasus5 Execution

- Once you have the input file and the volume grids are installed in the X\_DIR directory you can execute the code:
  - Serial version:  
`pegasus5 < peg. i`
  - MPI Parallel version using \$NCPUS cpus:  
`mpi exec -NP $NCPUS pegasus5mpi < peg. i`
- Note: mpi version requires that all CPUs have access to the same copy of the working directory



# Pegasus5 Output

- Pegasus5 creates a directory named **WORK** which contains all of the intermediate output files created by each internal task
  - Typically no need to examine or read these files directly
  - In order to re-run a case from the beginning, simply remove **WORK**
- All informational output written to a file named **log.mmdd.hhmm**. Examine this file to see what Pegasus5 did
  - Note: during execution the mpi version will create multiple log files which will be concatenated together upon successful completion of the run
    - log.mmdd.hhmm.0000
    - log.mmdd.hhmm.0001
    - log.mmdd.hhmm.0002
    - ...
    - log.mmdd.hhmm.0015
- The **XINTOUT** file contains all of the interpolation stencils and blanking information used by the flow solver



# Post Execution

- Examine log file and verify successful completion
- Examine minimum hole cuts and make sure no active points are left inside a solid body
  - Plot hole boundaries in plot3d
  - Plot grid slices in overgrid, tecplot, fieldview, etc
  - Look for orphan points left inside a solid body
- Examine and eliminate cause of orphan points



# End of Log File: Stencils and Orphans

Mesh Name	Interpolated Boundary Points	Interpolation Stencil	Orphan Points
fuselage	Level 1: 10634 Level 2: 24578 Total: 35212	Level 1: 32934 Level 2: 10498 Total: 43432	1st Fringe: 0 2nd Fringe: 0(Fixed) Total: 0
wing	Level 1: 38609 Level 2: 49279 Total: 87888	Level 1: 30486 Level 2: 12261 Total: 42747	1st Fringe: 2 2nd Fringe: 0(Fixed) Total: 2
wingcap	Level 1: 20251 Level 2: 242 Total: 20493	Level 1: 12491 Level 2: 22748 Total: 35239	1st Fringe: 0 2nd Fringe: 0(Fixed) Total: 0
fuselagefillet	Level 1: 8827 Level 2: 14321 Total: 23148	Level 1: 9584 Level 2: 13172 Total: 22756	1st Fringe: 2 2nd Fringe: 0(Fixed) Total: 2
Grand Total	Level 1: 262641 Level 2: 267467 Total: 530108	Level 1: 262641 Level 2: 267467 Total: 530108	1st Fringe: 14 2nd Fringe: 0 Total: 14



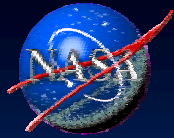
# End of log file: Execution Time

PROCESS	CPU(sec)	WALL(sec)	Sub-procs	Max sub-proc(sec)
proj ecti on	13. 875	2. 328	122	1. 672
adt	4. 656	1. 266	13	0. 906
i nterpol ate	65. 922	24. 586	122	3. 867
auto_hbound	66. 438	26. 898	3	26. 906
man_hbound	0. 000	0. 000	0	0. 000
auto_cut	42. 234	4. 906	30	4. 805
man_cut	0. 000	0. 000	0	0. 000
comp_hol e	1. 156	0. 141	13	0. 125
spec_i nt1	0. 508	0. 055	13	0. 062
spec_l evel 1	8. 078	0. 859	13	0. 867
l evel 1fi x	1. 734	2. 305	1	1. 734
spec_i nt2	19. 859	2. 195	61	0. 930
spec_l evel 2	9. 859	1. 023	13	1. 016
xi ntout	1. 469	1. 477	1	1. 469

SUM of PROCESS TIME for all processes (secs): 235. 789

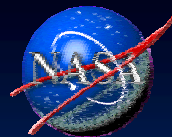
ELAPSED WALL TIME(secs): 37. 703

EXECUTION SPEED-UP = 6. 25 usi ng 15 processors.

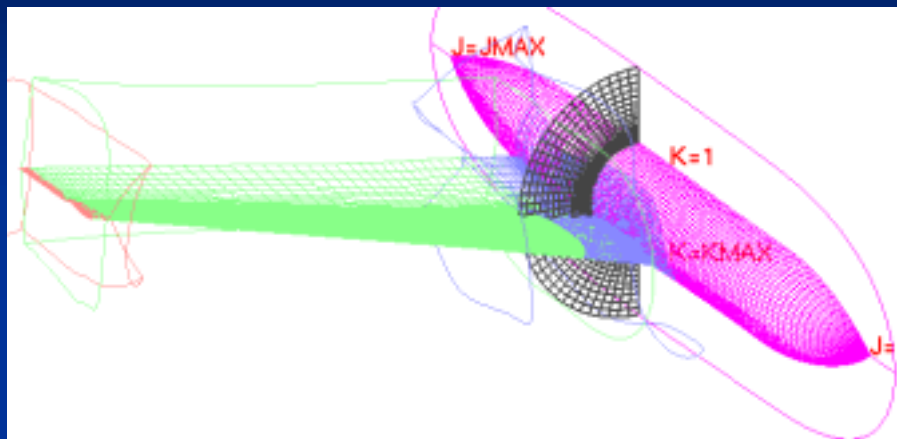


# Output: `peg_plot`

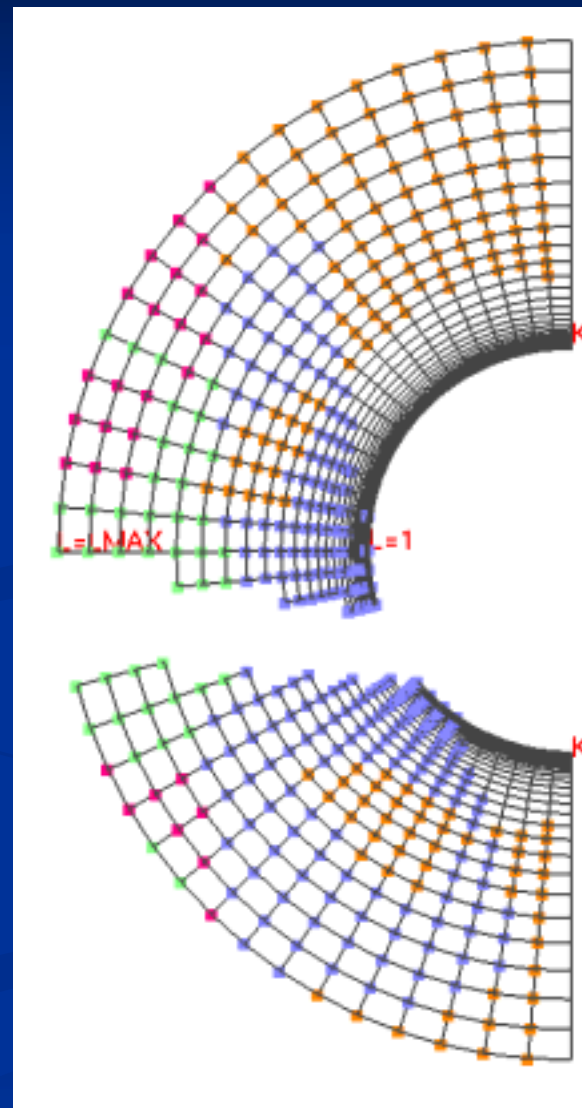
- Grid file: use the `peg_plot` program to create the grid file used by the flow solver, and to plot and check the results of the Pegasus5 run
  - Use `peg_plot` option 3 first to examine the results of the hole cutting
  - The `peg_plot` options 1 and 2 blank out the higher-level fringes in the resulting grid file
    - This illustrates the borders of where information is passed between overlapping zones
    - Useful when plotting the flow solution as it minimizes the overlap
- Note: Overflow does not use the `iblack` array in the grid file, so any `peg_plot` option works when creating the grid file that will be passed to Overflow

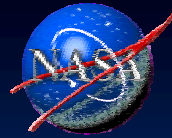


# Example: `peg_plot 3`

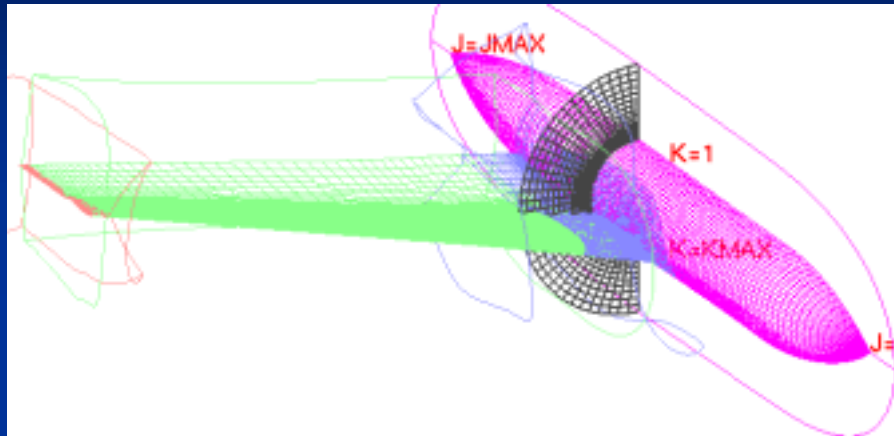


- Wing-body example using `peg_plot` option 3
- View the fuselage zone in `overgrid`
- Shows auto hole cut by the wing
- Fringe points shown with colored symbols

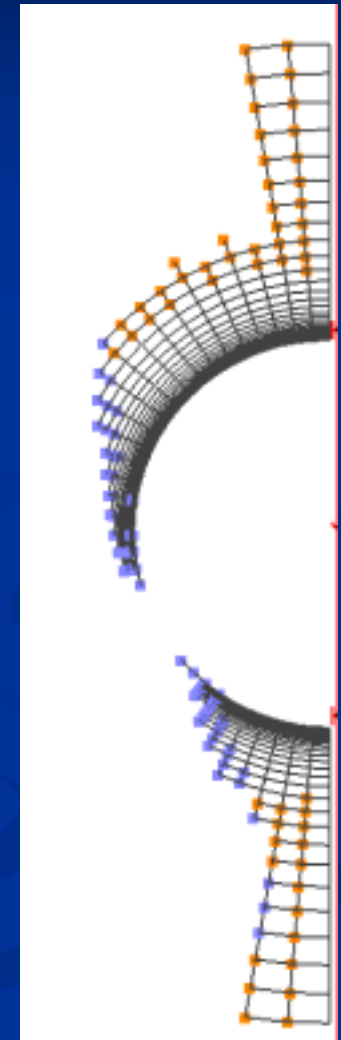


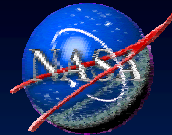


# Example: `peg_plot 2`



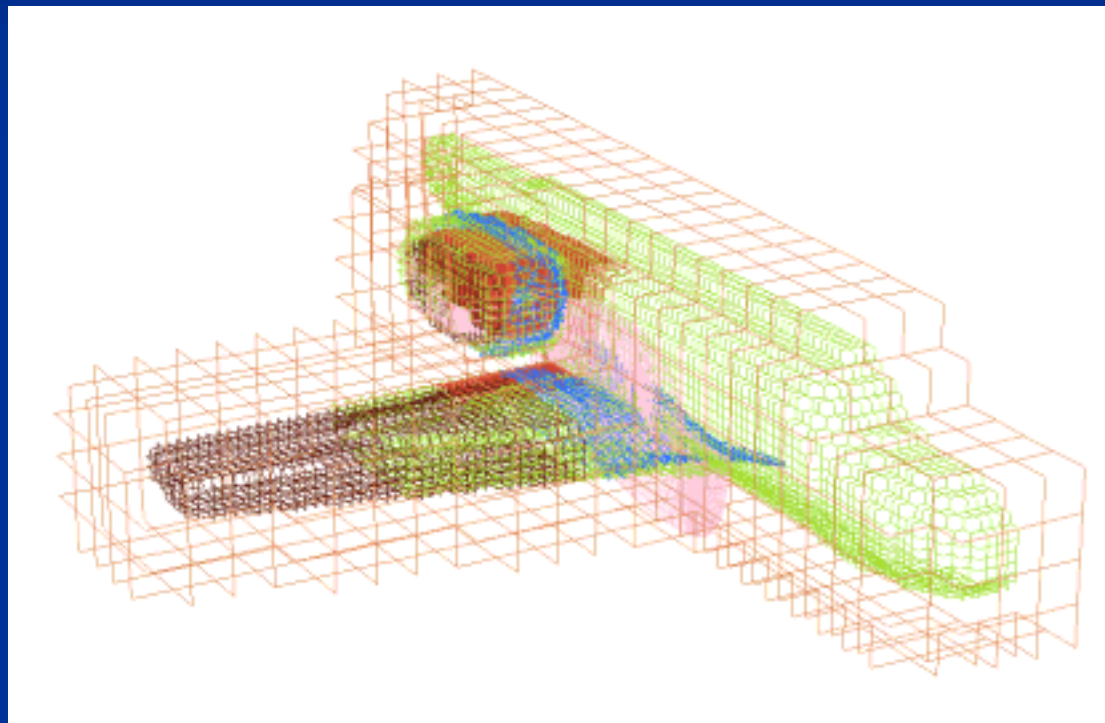
- Wing-body example using `peg_plot` option 2
- Higher-level fringe points have been blanked out
- Shows the virtual overlap after the Level-2 interpolation
- Flow-solver still keeps the higher-level fringes active: they can be used as donor cells for other zones





# Examining the Hole Cuts

- Use `plot3d` function  
2: plots the outlines of  
the holes
- Use `Overgrid`, etc:  
plot slices through  
grids
- Search log file for  
“composite hole”:  
lists number blanked  
points in each mesh
- Use `peg_hole_surf`  
to extract grid  
surfaces used by each  
\$HCUT auto hole cutter





# Custom Hole Cutting

- The **\$HCUT** namelists are used to define separate auto hole-cutters
  - By default, no \$HCUT namelist is included in input file, and pegasus5 uses ALL solid-wall surfaces to cut holes from ALL zones
  - Adding an \$HCUT entry eliminates this

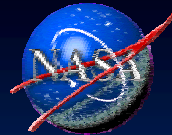
Solid walls of these zones must form a fully enclosed volume

List of zones which are cut

```
$HCUT NAME = 'hcutter1',  
        MEMBER = 'body1', 'body2',  
        INCLUDE = 'bodynose', 'wing',  
        wingcol',  
        CNX = 512, CNY = 512, CNZ = 512,  
        CARTX = -100.0, 100.0,  
        CARTY = -50.0, 50.0,  
        CARTZ = 0.0, 100.0,  
        $END
```

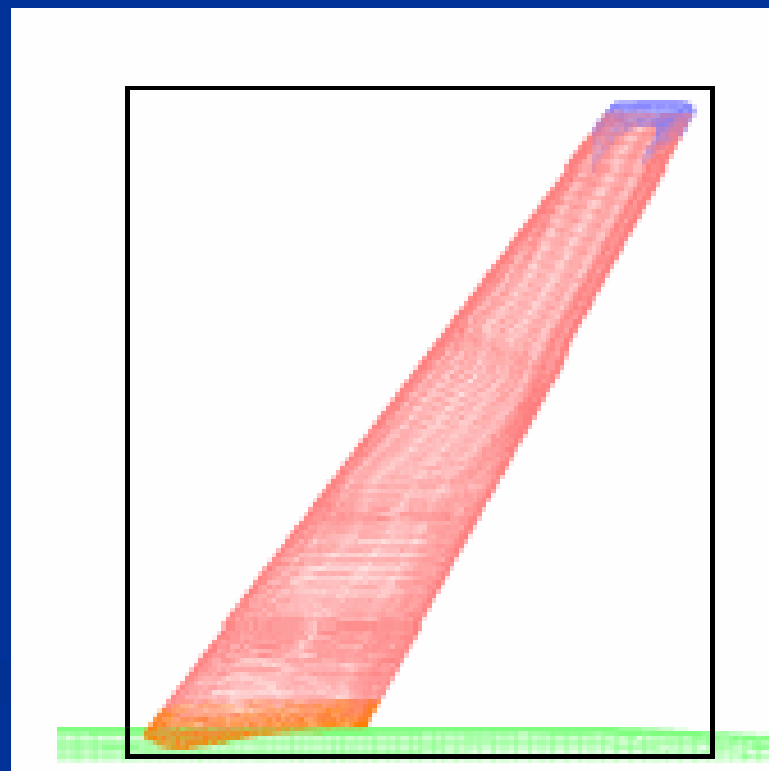
Cartesian map dimensions

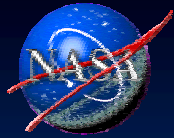
Bounding box of hole cutter



# Custom Hole Cutting

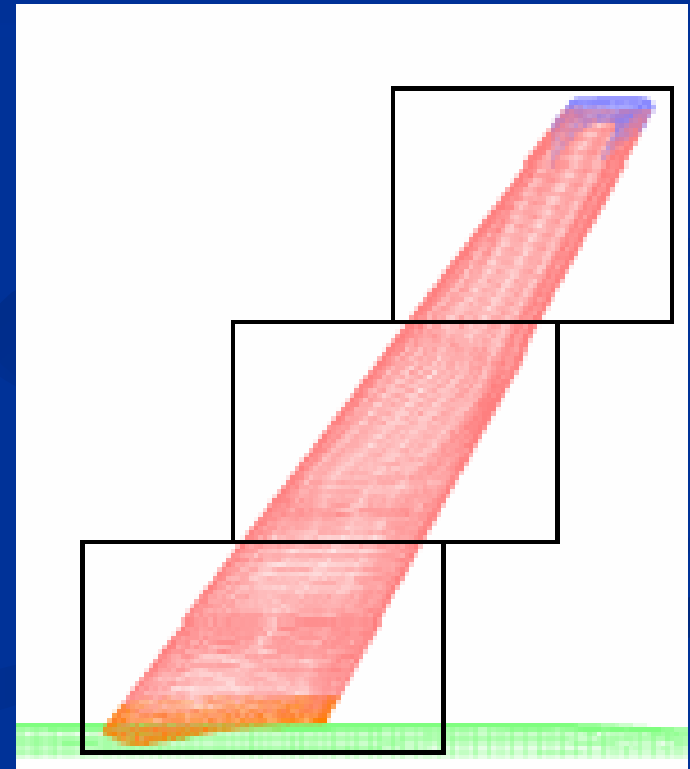
```
$HCUT NAME = 'wi nghol e' ,  
  MEMBER  = 'wi ng' , 'wi ngcol' , 'wi ngcap' , ' body' ,  
  INCLUDE = ' body' , 'wi ngbox' , ' bodybox' , ' farbox' ,  
  CARTX   = 100.0, 400.0,  
  CARTY   = 10.0, 150.0,  
$END
```

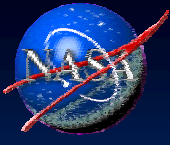




# Custom Hole Cutting

```
$HCUT NAME = 'winghole1',  
  MEMBER = 'wing', 'wingcol', 'body',  
  INCLUDE = 'body', 'wingbox', 'bodybox', 'farbox',  
  CARTX = 100.0, 250.0,  
  CARTY = 10.0, 51.0,  
  $END  
$HCUT NAME = 'winghole2',  
  MEMBER = 'wing',  
  INCLUDE = 'wingbox', 'farbox',  
  CARTX = 200.0, 350.0,  
  CARTY = 50.0, 101.0,  
  $END  
$HCUT NAME = 'winghole3',  
  MEMBER = 'wing', 'wingcap',  
  INCLUDE = 'wingbox', 'farbox',  
  CARTX = 240.0, 400.0,  
  CARTY = 100.0, 150.0,  
  $END
```





# Hole-Cutting Troubleshooting

- No holes cut due to leak or gap in solid-wall surfaces
  - Use **CARTX**, **CARTY**, **CARTZ** to seal gap
  - Use PHANTOM zone to seal gap
  - Edit input file and extend solid-wall boundary
- Holes too small near thin bodies, such as TE of a thin wing:
  - Increase **OFFSET** to enlarge holes
  - Increase **CNX**, **CNY**, **CNZ** to improve resolution of Cartesian map
- Hole points not cut properly near collar grids
  - Increase **OFFSET** to enlarge holes
- Holes cut at zone edges adjacent to solid walls in regions of high curvature – can occur with inadequate resolution relative to curvature
  - Increase grid resolution
  - Use **\$REGION** and **\$VOLUME** namelists to “unblank” holes



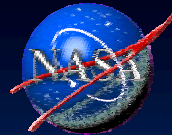
# Manual Hole Cutting

- Manual hole-cutting functionality from pegasus4 has been retained in Pegasus5
  - Can be used as an additional tool to refine holes
- **\$BOUNDARY/\$SURFACE** namelists
  - Can specify a group of zonal surfaces which will cut holes in the specified zones
- **\$BOUNDARY/\$BOX** namelists
  - Can specify range of x,y,z coordinates of a box which will cut holes in the specified zones
- **\$REGION/\$VOLUME** namelists
  - Can specify range of j,k,l zonal indices to create a hole or to unblank part of an existing hole



# Orphan Points

- Orphan points are fringe points for which no valid interpolation donor can be found
- Orphans are reported in the log file, in the output of `peg_plot`, and using the `peg_orph` program
- 2<sup>nd</sup>-level fringe-point orphans are reset to active interior points
- Overflow will update the solution data at orphan points by averaging the neighboring grid points
  - A few isolated orphan points are usually acceptable, but it is advisable to find and fix most or all orphans
  - Orphans on solid-wall surfaces usually indicate a serious problem with surface resolution or projection, and should be fixed
- Plot orphans using the `plot3d` or `overgrid` programs



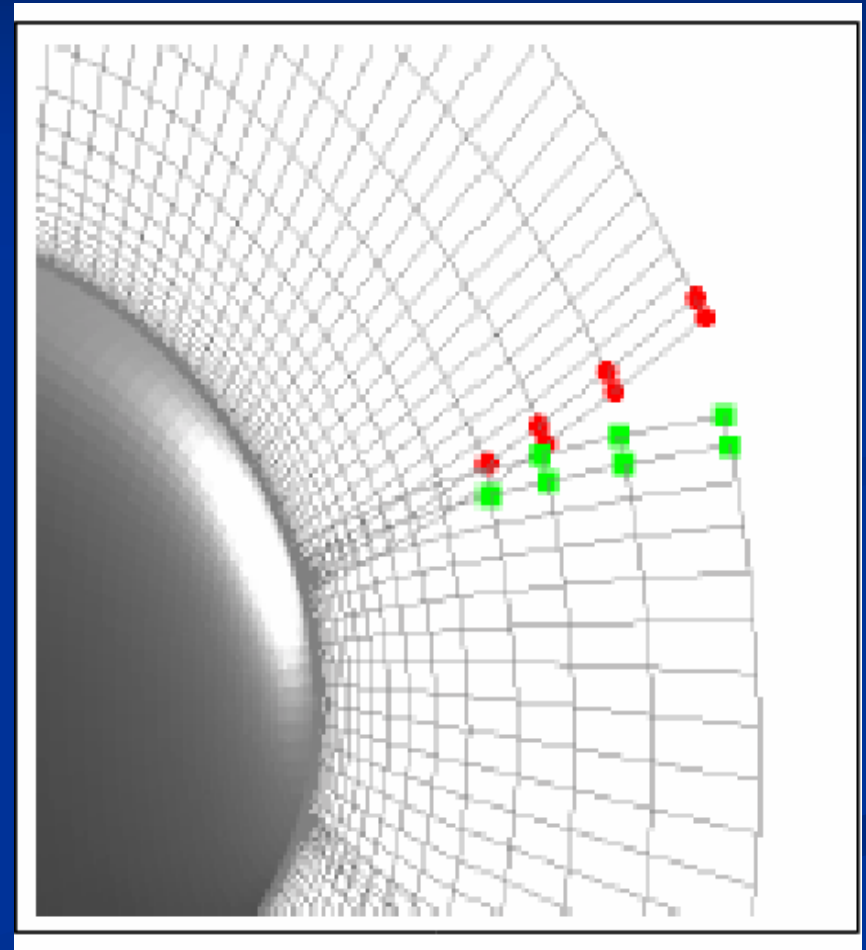
# Causes of Orphans

- Bad hole cuts
- Insufficient overlap
- Poorly resolved geometry in regions of surface curvature
- Inappropriate or missing boundary conditions



# Insufficient Overlap

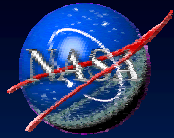
- Possible fixes:
  - Increase surface-grid overlap
  - Add more splaying to the boundary conditions in *hypgen*
  - Add a cartesian box grid to resolve the open space





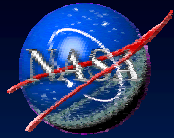
# Utility Codes

- `peg_setup`: creates
- `peg_hole_surf`: creates plot3d grid files containing surfaces of each \$HCUT entry
- `peg_plot`: creates composite plot3d grid file
- `peg_diag`: produces diagnostic file for plotting interpolation parameters and connection info
- `peg_orph`: lists orphan points for each zone
- `peg_proj`: creates diagnostic plot3d files for projection
- `XINTegrity`: verifies integrity of the XINTOUT file



# Summary

- Pegasus5 successfully automates most of the oversetting process
  - Dramatic reduction in user input over previous generations of overset software
  - Reduced complex-geometry oversetting time from weeks to hours
  - Significant reduction in user-expertise requirements
- Not a “black-box” procedure: care must be taken to examine the resulting grid system
- Additional documentation and examples available online:  
<http://people.nasa.gov/~rogers/pegasus/status.html>



# Nomenclature

- **Grid System:** A collection of zones together with boundary conditions and connectivity data ready to input into a flow solver
- **Zone:** a single structured grid composed of ordered grid points
- **Cell:** a hexahedron composed of 8 grid points and 6 faces
- **Grid point:** a single point in a zone identified uniquely by its j,k,l indices
- **Fringe point:** a grid point which will be updated in the flow-solver via interpolation of data from a neighboring zone
- **Outer-boundary fringe point:** a fringe point on the boundary edge of a zone
- **Hole-boundary fringe point:** a fringe point adjacent to a hole point
- **Hole point:** a grid point which has said to have been “blanked out” and whose data will not be used by the flow solver
- **Orphan point:** a fringe point for which a valid donor cell cannot be found
- **Interior point:** a grid point which does not lie on the zonal boundary
- **Iblank value:** each grid point is assigned an integer value to denote type of point:
  - iblank<0: fringe point
  - iblank=0: hole point
  - iblank=1: active interior point
  - iblank=101: orphan point